

Some Challenging Problems in (Active) Flow and Noise Control *A (Limited) Experimental Perspective*

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***Future Directions in CFD Research
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Research Sponsors

*AFOSR, AFRL, DARPA, BOEING NASA,
ONR, State of Florida and others*

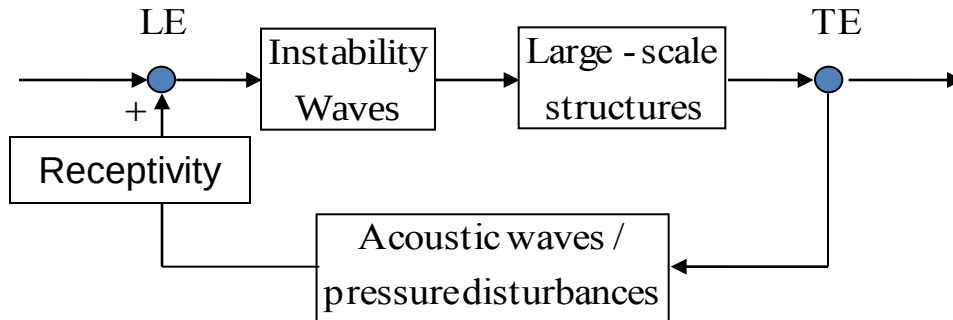


- **Problems**
 - Supersonic Cavity Flows
 - Supersonic Impinging Jets
 - Separated Flows
- **Experimental & Computational Results**
 - Base Flowfield
 - Response to Active Control
 - Steady Microjets
 - Pulsed Microjet Actuators
- **Final Thoughts**



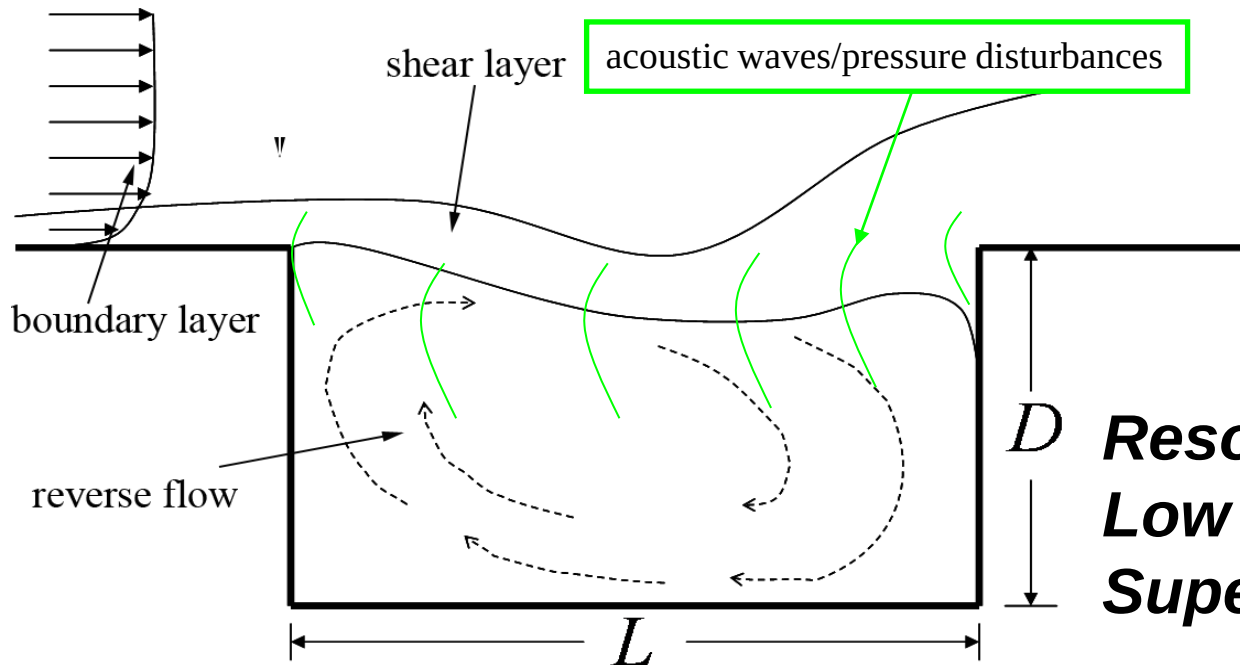
Cavity Flows

Flow –Acoustic Resonance



**Modified Rossiter's Model
(Heller & Bliss, 1975)**

$$St = \frac{fU_{\infty}}{L} = \frac{(m-r)}{\left(M \left(1 + \frac{\gamma-1}{2} M^2 \right)^{-0.5} + \frac{1}{k} \right)}$$



m – Mode no.
 r – Phase delay
 k – $U_{\text{conv}}/U_{\infty}$

**Resonance observed for
Low subsonic to high
Supersonic Flows**

Cavity Flow Visualizations

Subsonic Flow

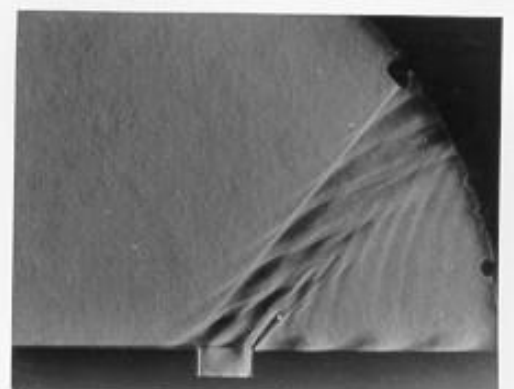
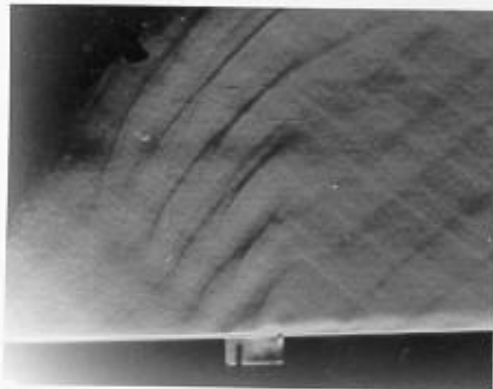
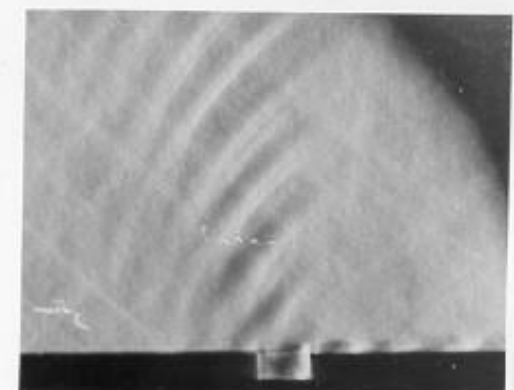
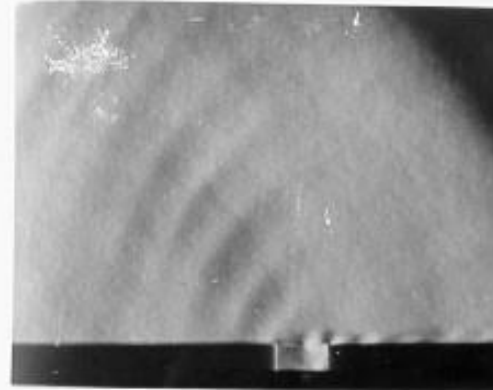


Phase-locked Schlieren Imag

$L/D = 2, M = 0.5$

(Kegerise, et al., 1999)

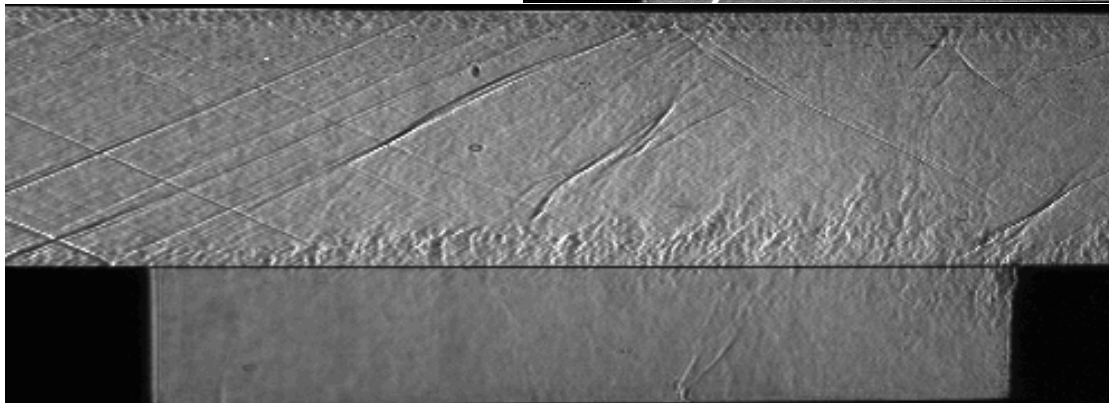
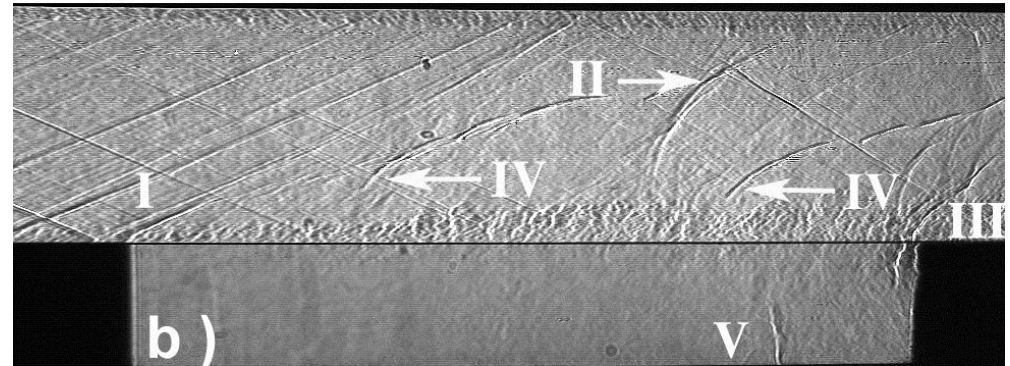
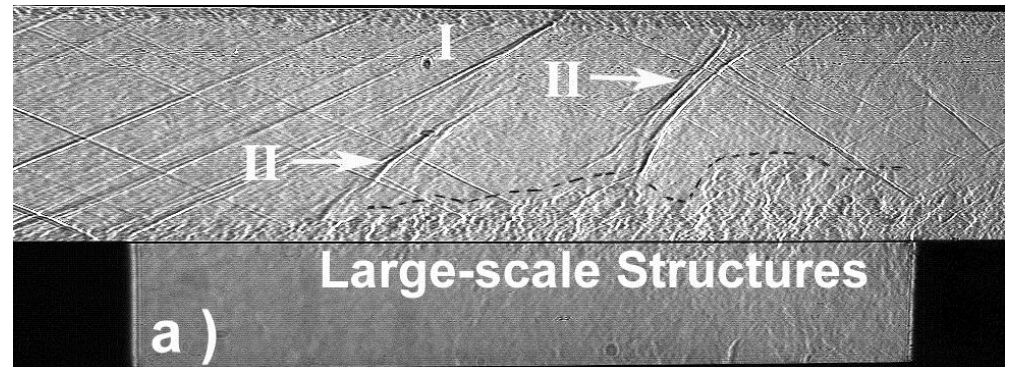
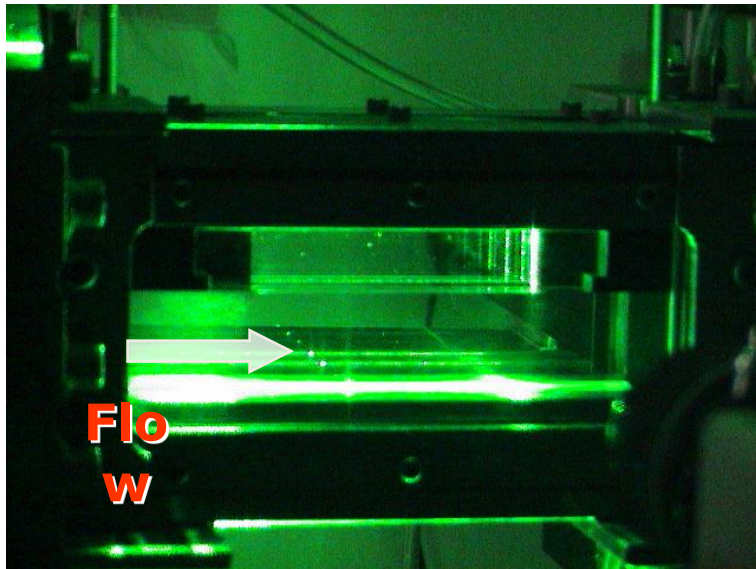
Subsonic to Transonic



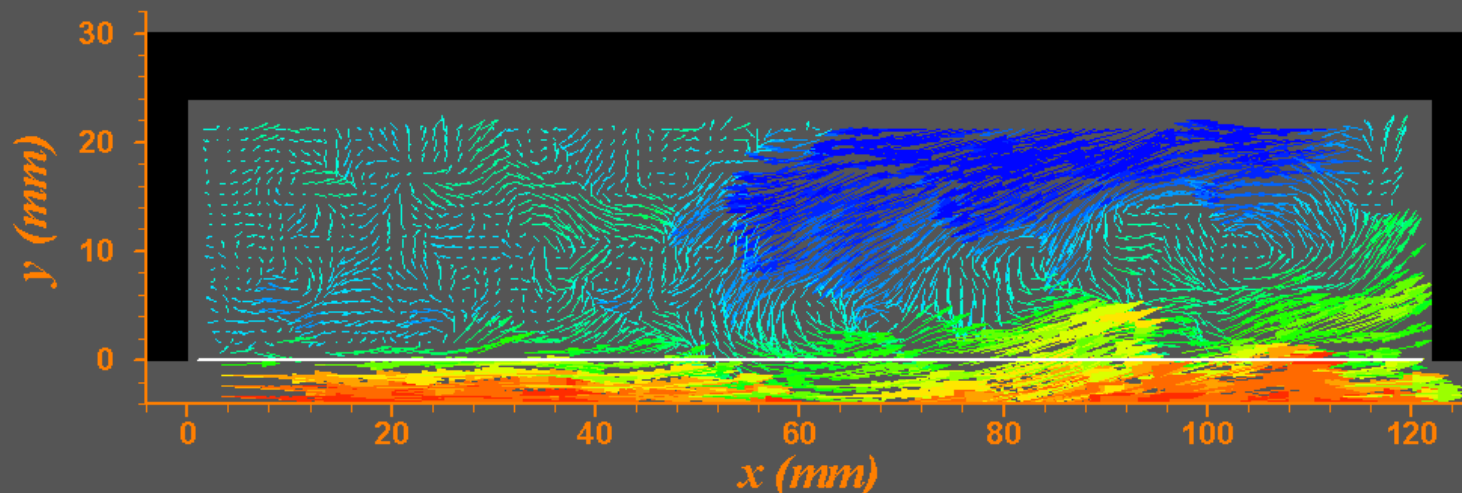
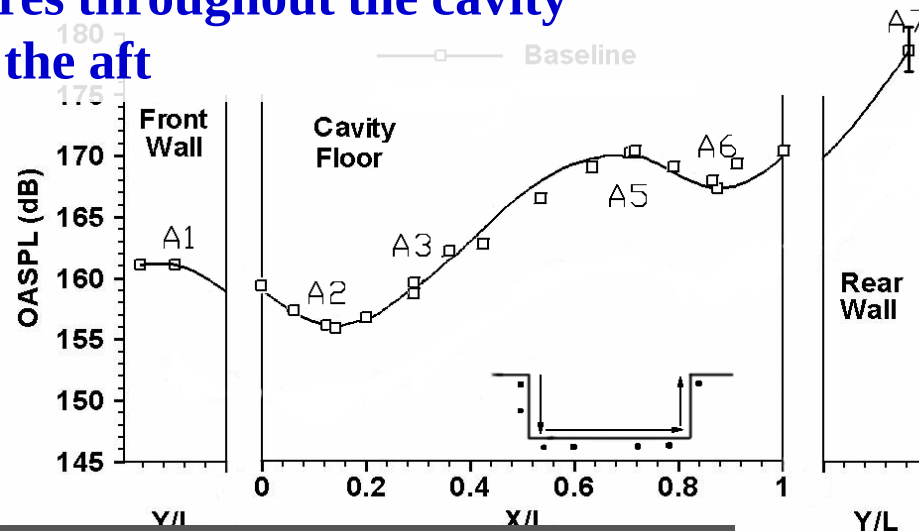
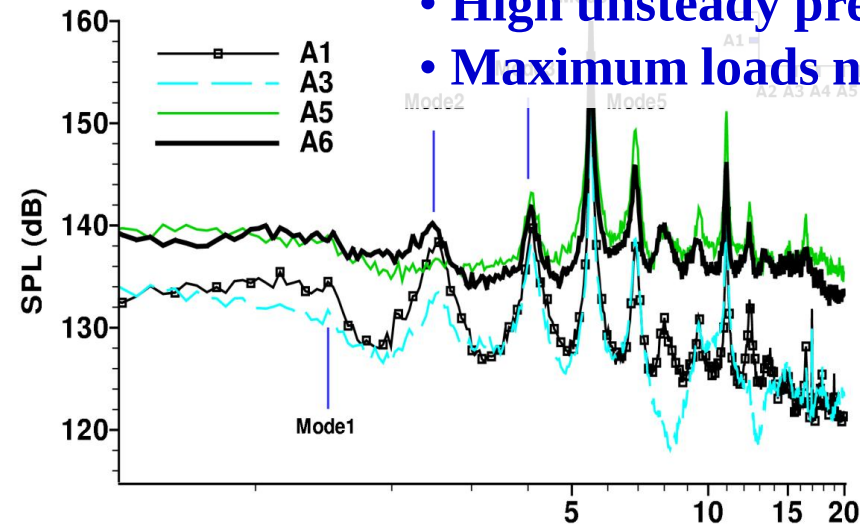
Krishnamurti, 1955

Supersonic Cavity Flow

$M = 2$

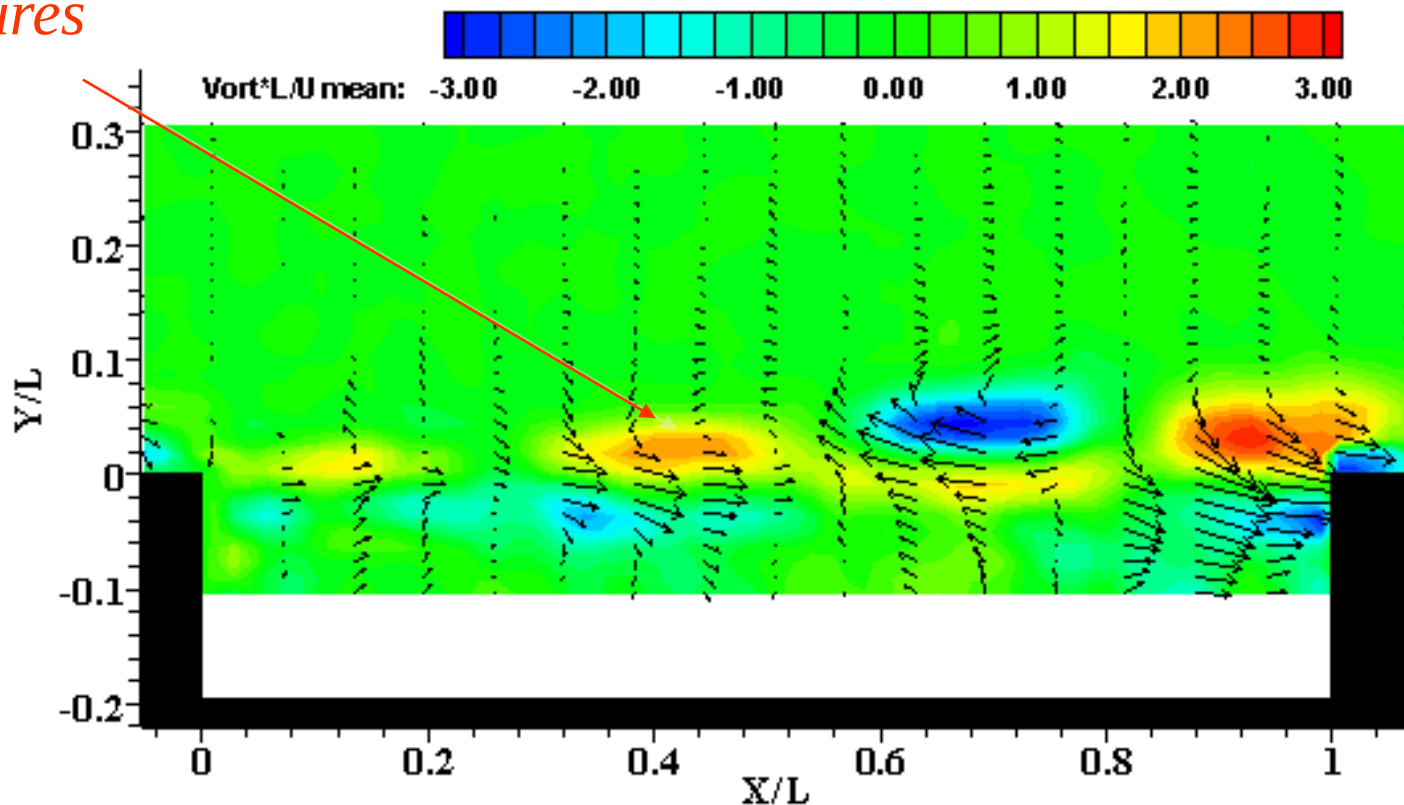


- High unsteady pressures throughout the cavity
- Maximum loads near the aft



Large- scale
structures

$L/D \sim 5$



$$\langle \omega \rangle = \omega - \bar{\omega}$$

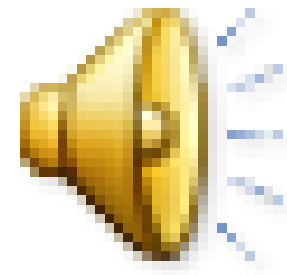
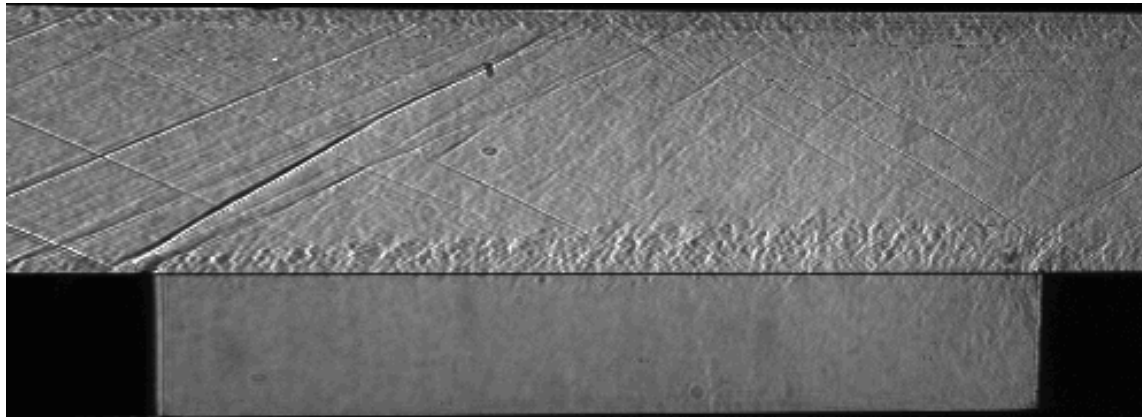
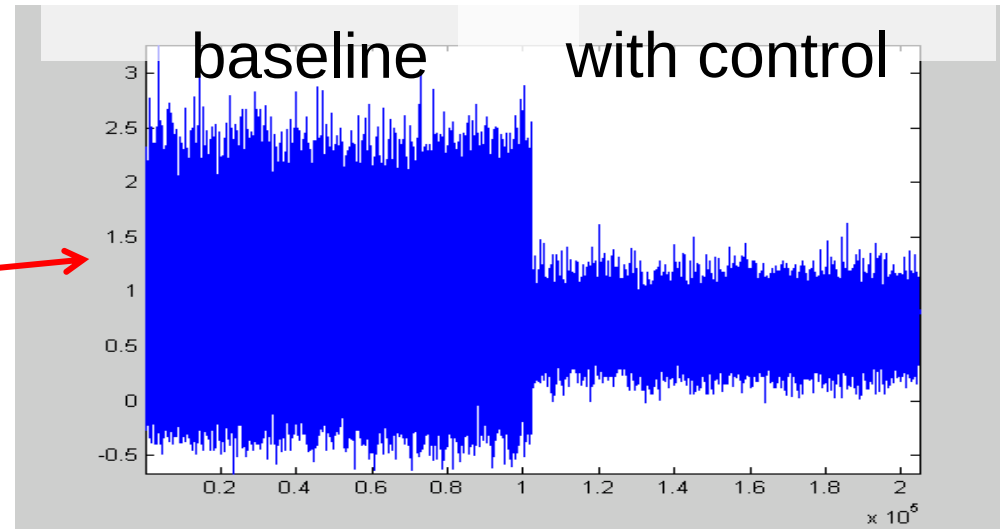
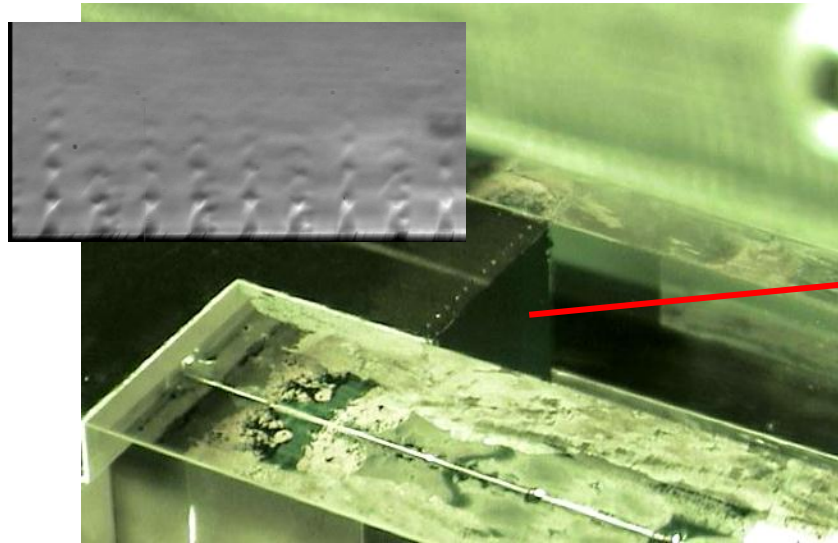
Periodic component

Phase-conditioned component

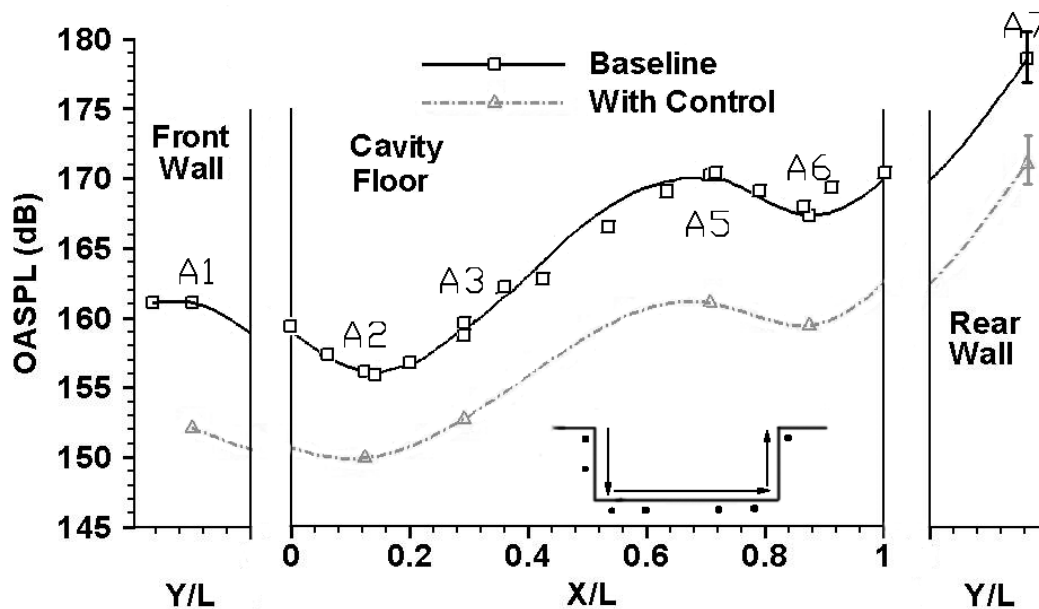
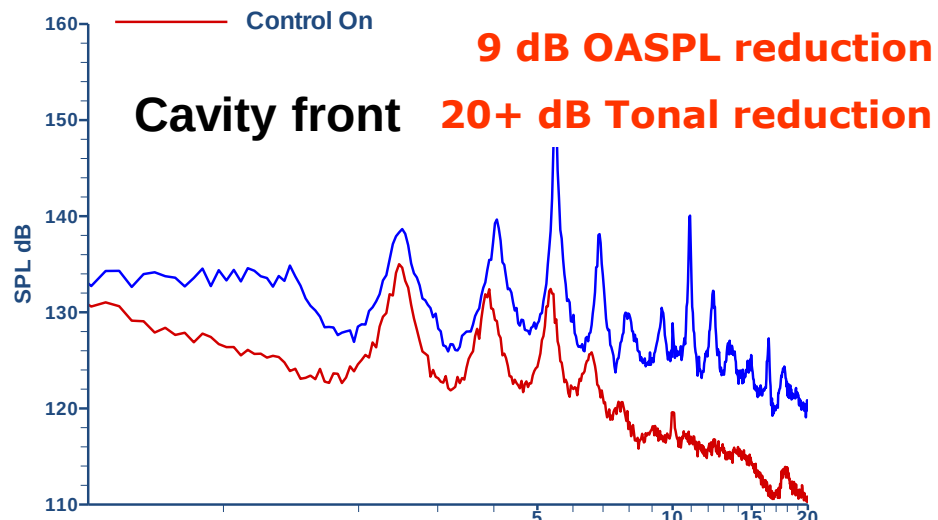
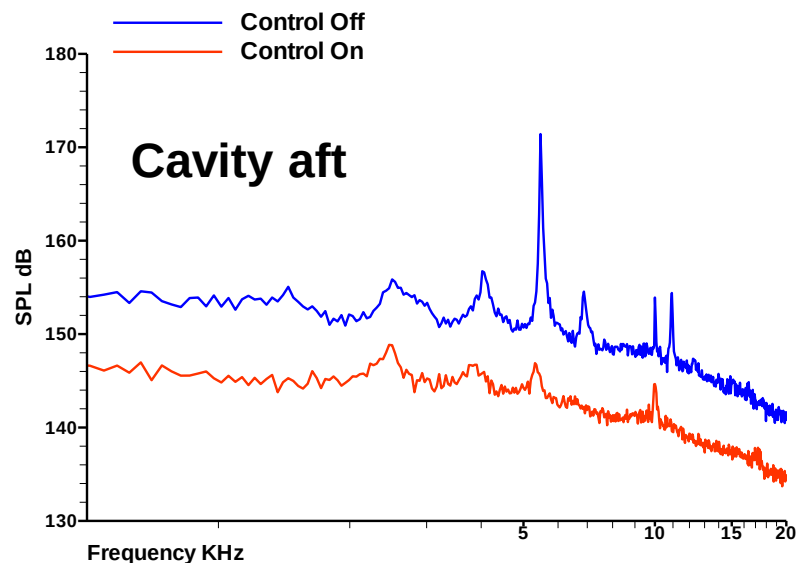
Ensemble averaged component

Mach 2 Cavity Flow

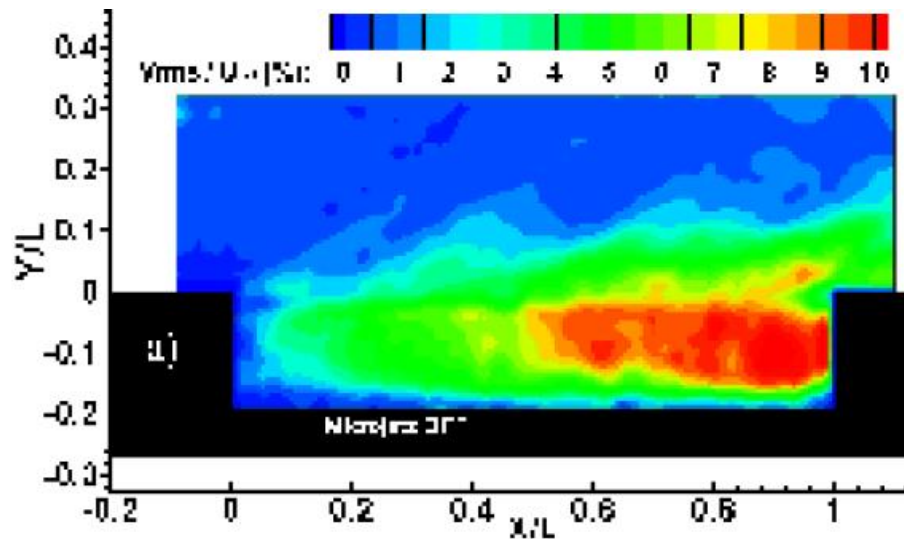
Effect of *Microjet Control*



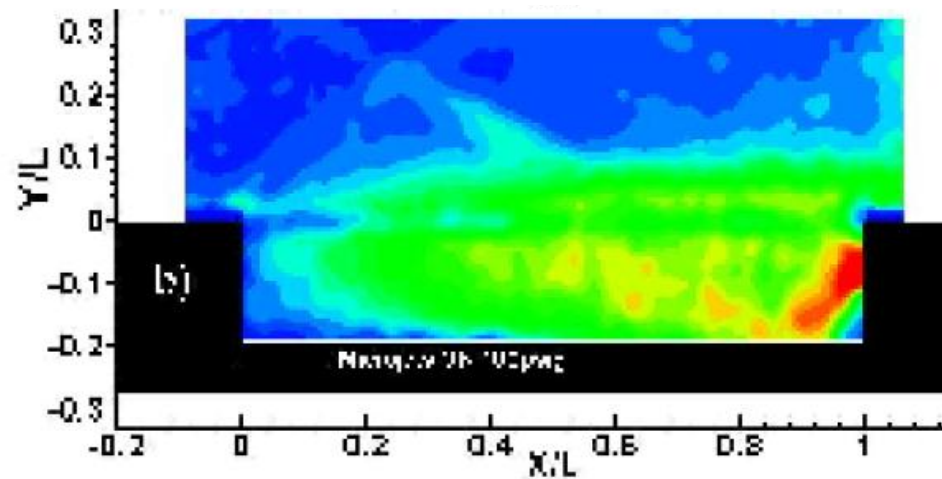
Effect of *Microjet Control* Unsteady Pressure Spectra



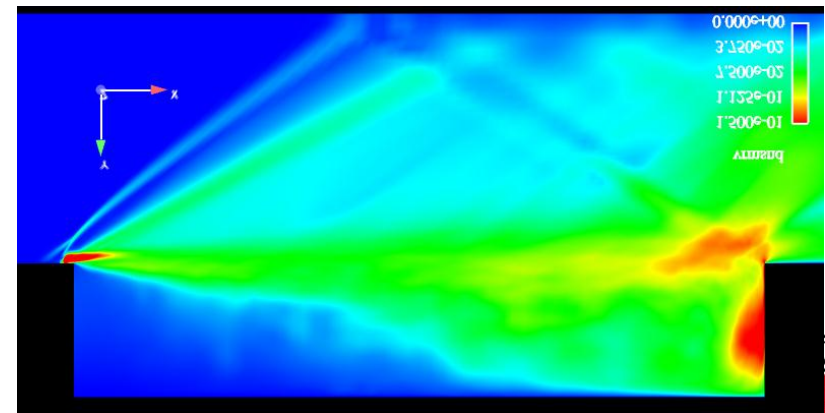
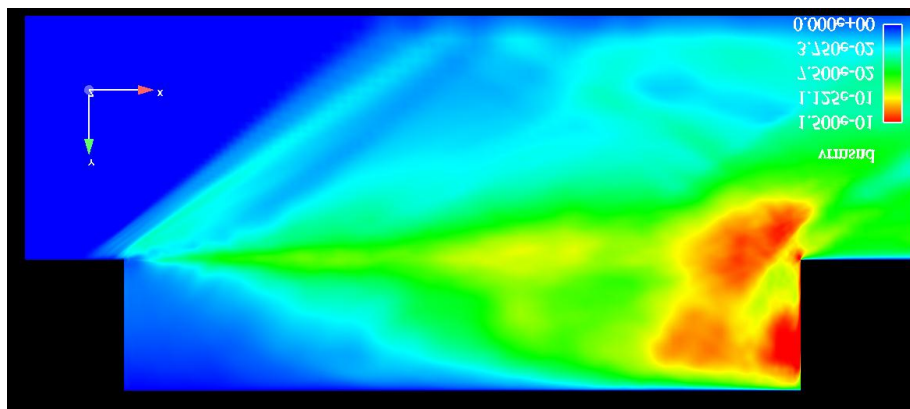
Ensemble-Averaged (using PIV)



Baseline



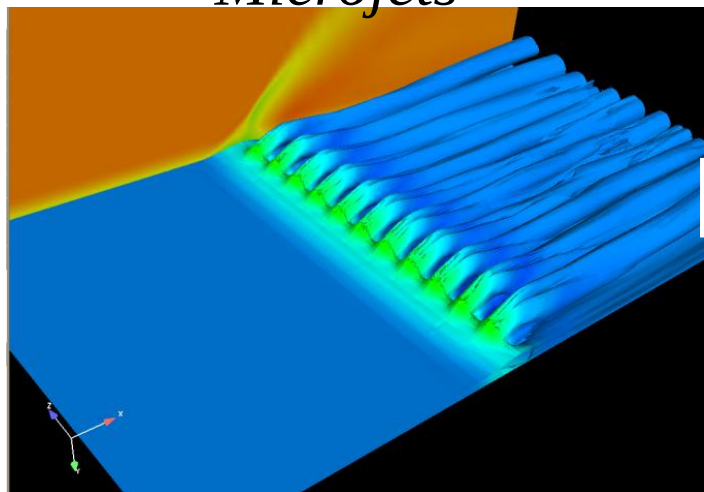
Microjet Control



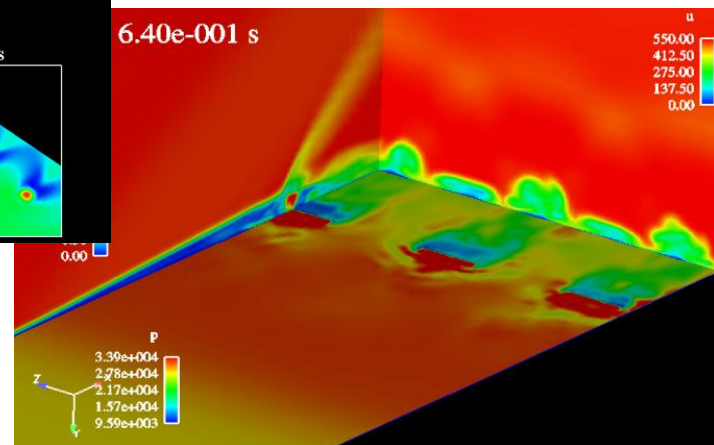
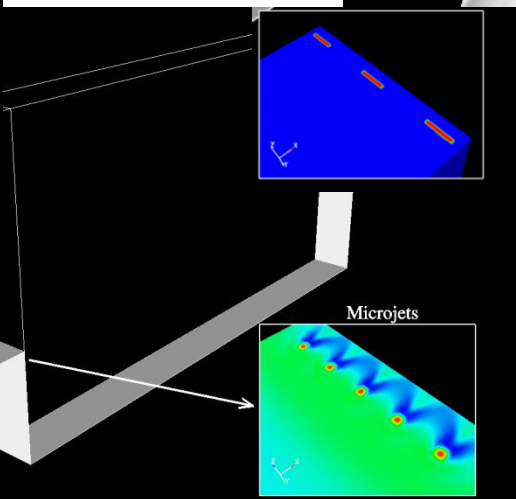
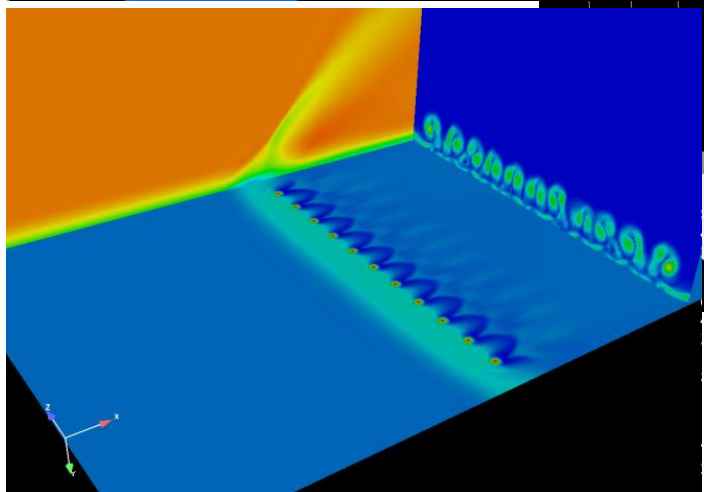
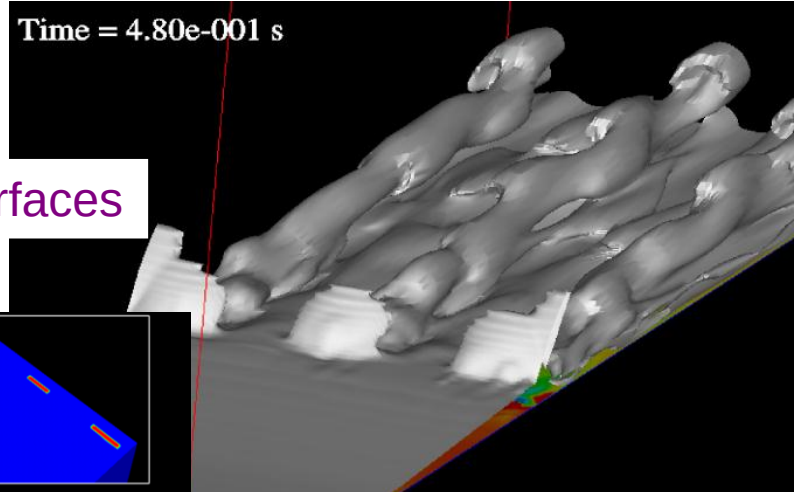
LES (courtesy CRAFT Tech)

Microjets & Slot Jets Simulations*

Microjets



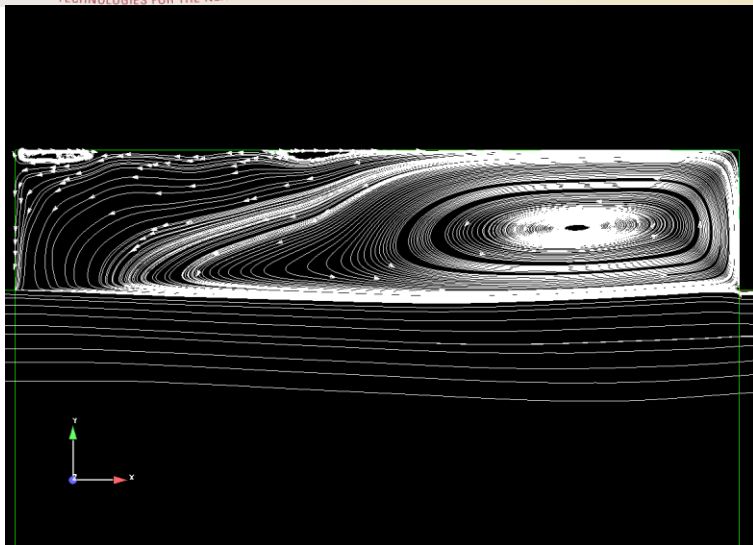
Velocity Iso-surfaces



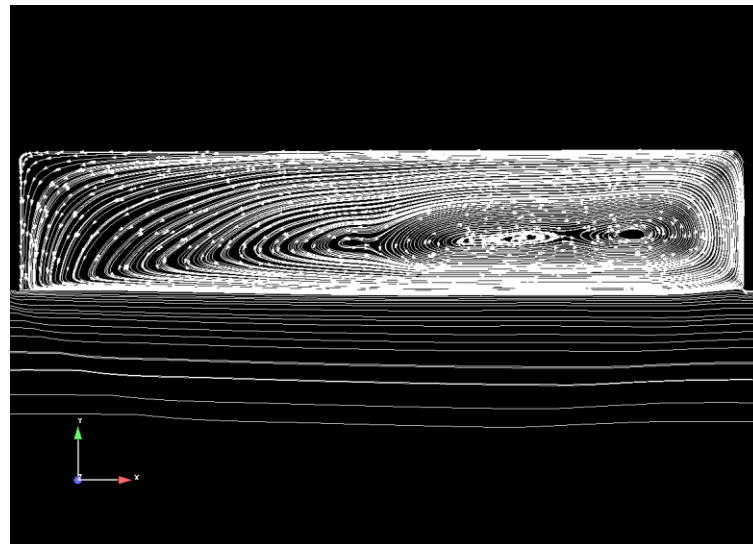
Slotjets

* S. Arunajatesan et al.
AIAA Jrnl, 2009

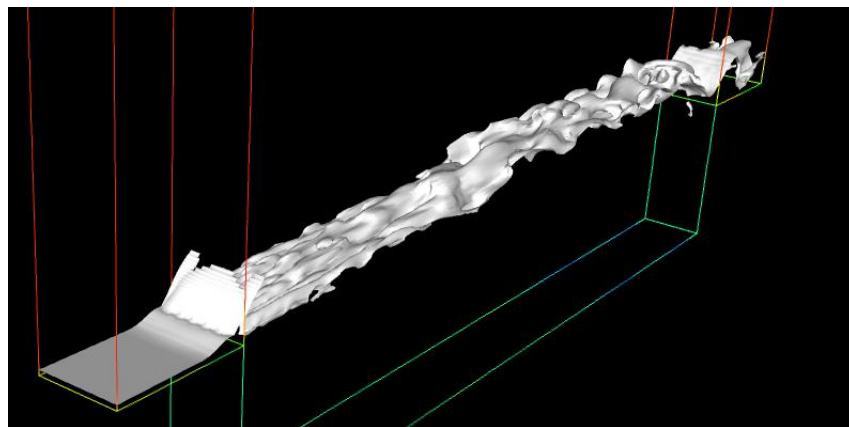
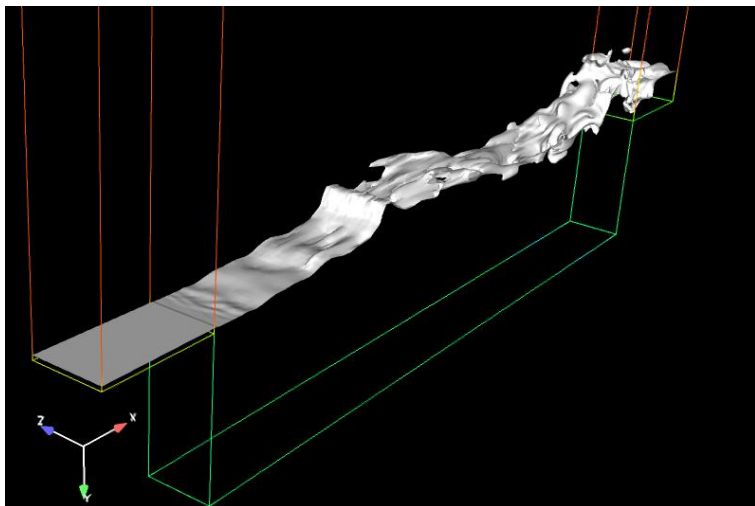
Effect of Microjets : Simulations*



Baseline



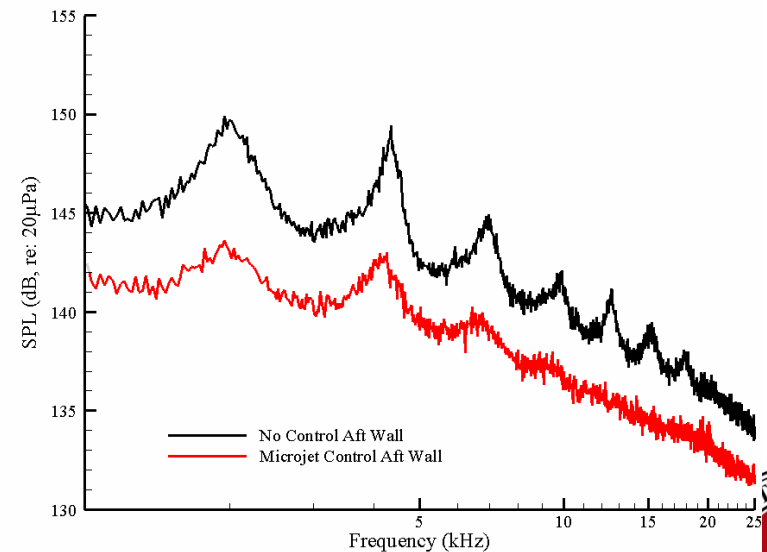
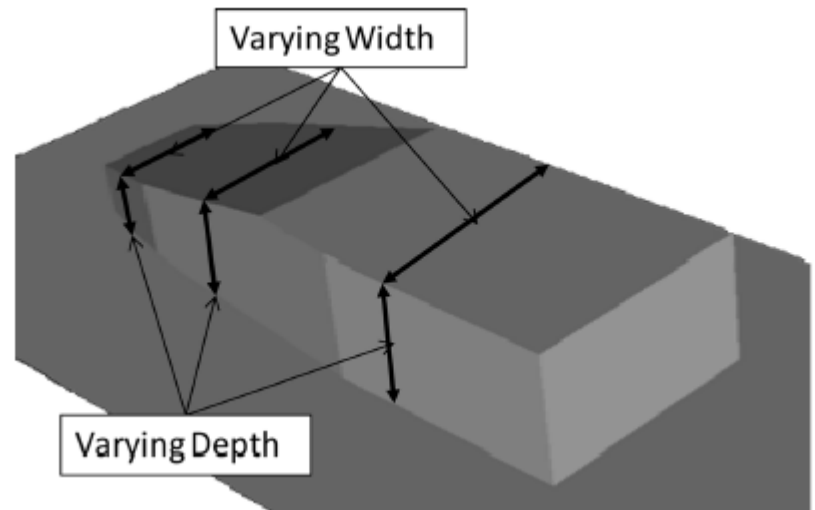
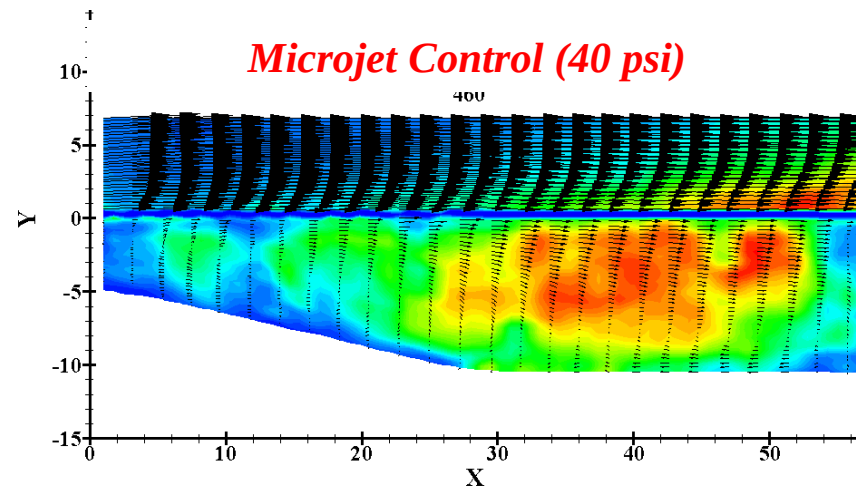
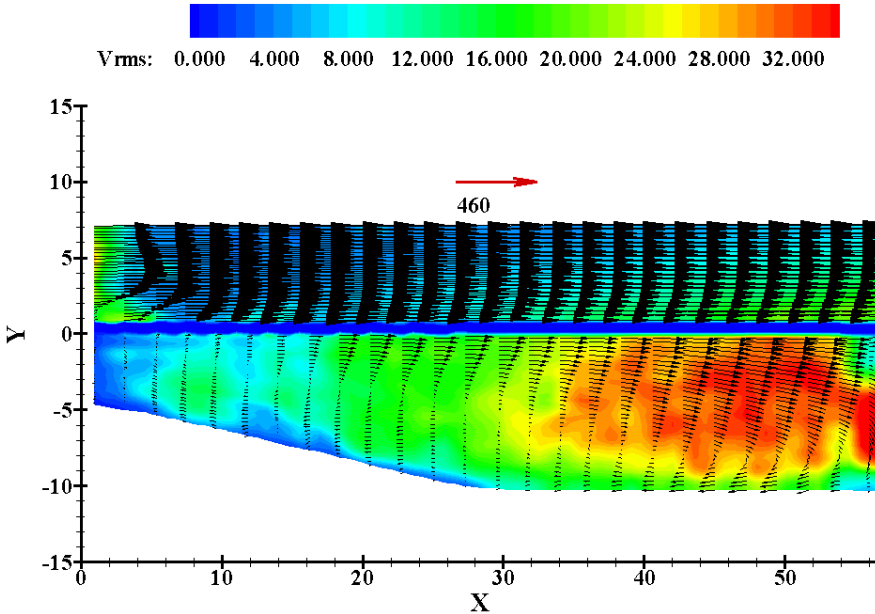
Microjet Control

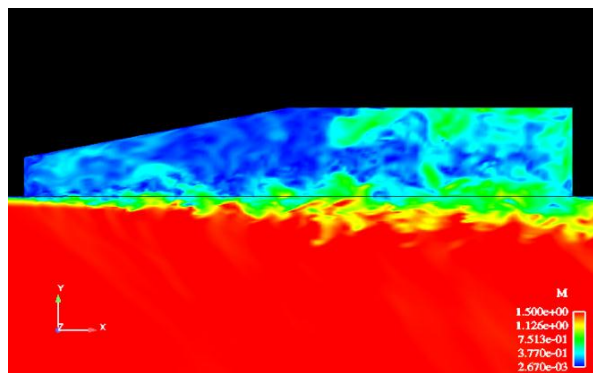


Velocity Iso-surfaces

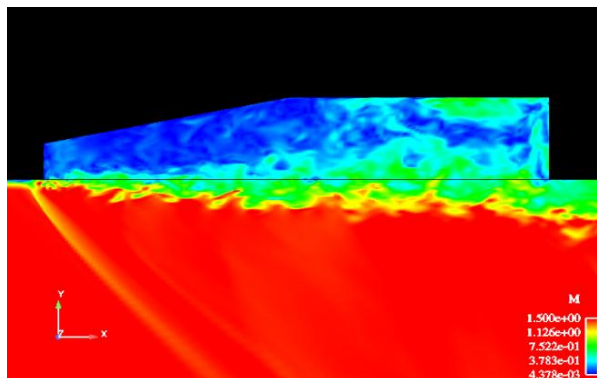
*Courtesy CRAFT Tech

Complex Cavity Flows

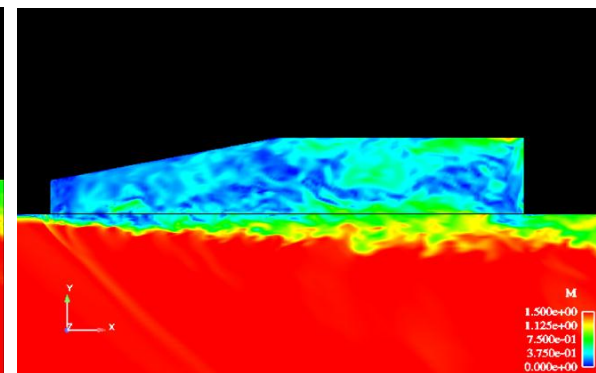




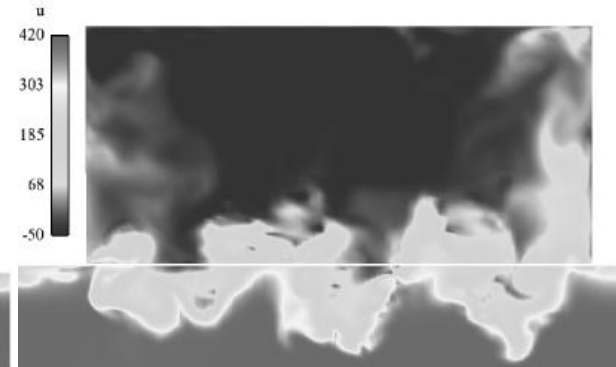
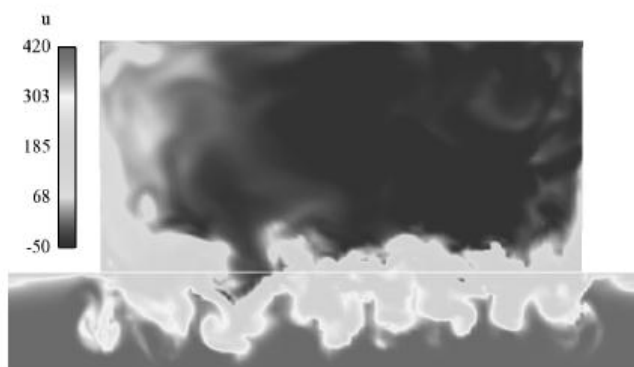
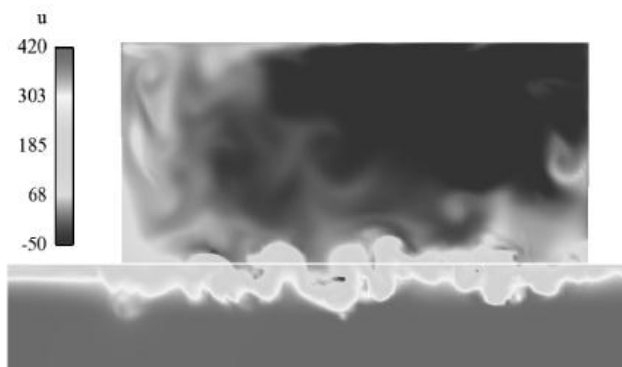
Baseline



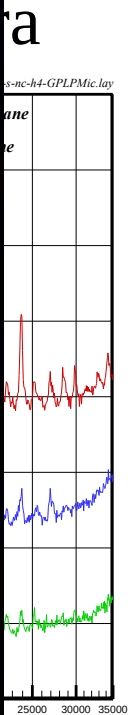
Microjets



Slotjets



Simulations provide insight into the 3-D nature of the flow

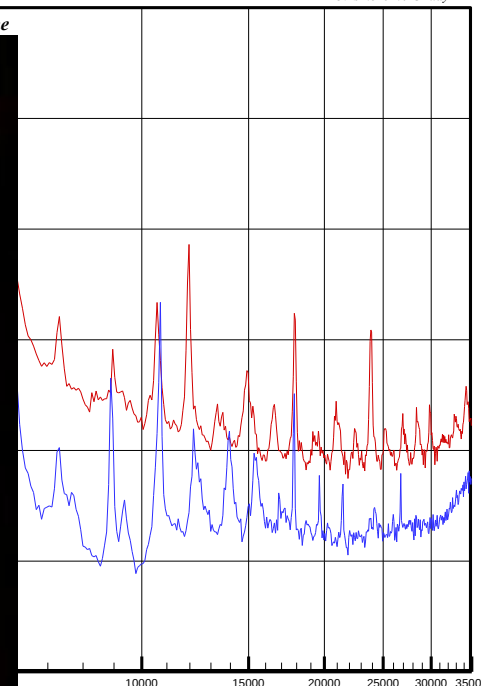


Control of Impinging Jets Using *Steady* Microjets

190

NPR 3.7 $h = 4.0$ Ground Plane

37-s-nc10-h4-GP.lay



Frequency(Hz)

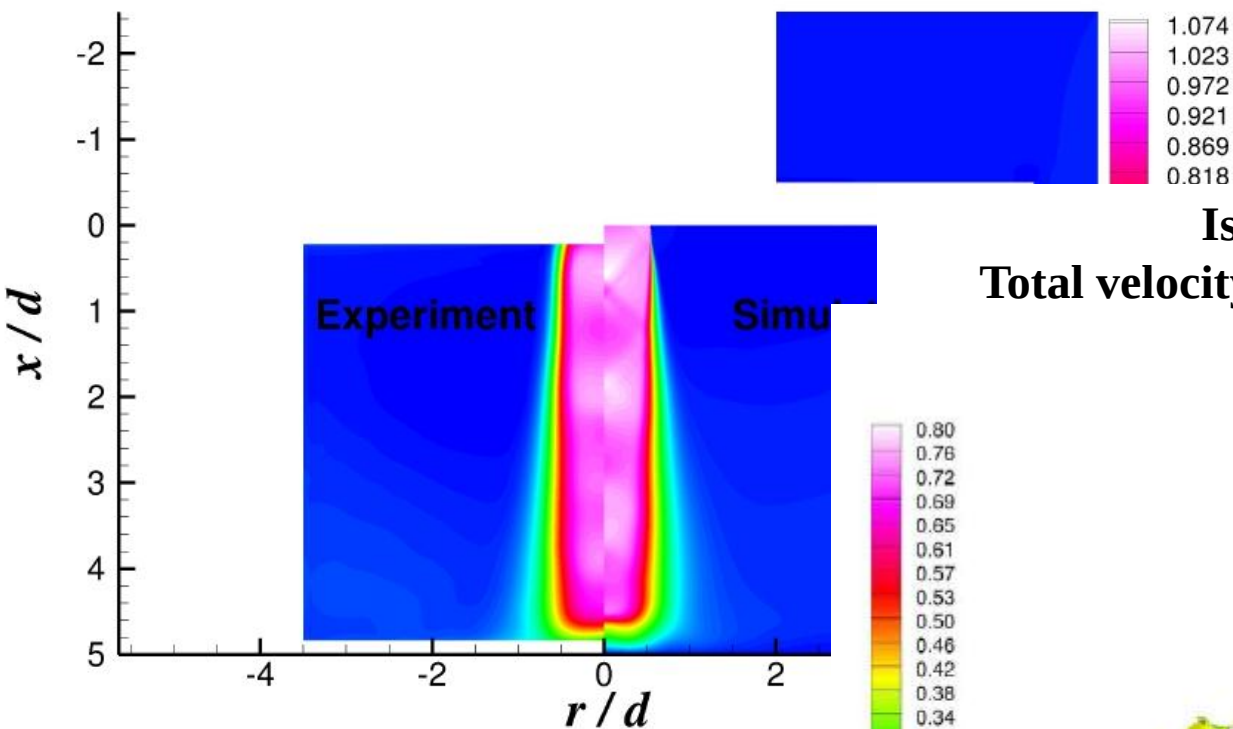
pressures and Noise
d by 4-12+ dB

To date, control demonstrated for cold and hot
impinging Jets

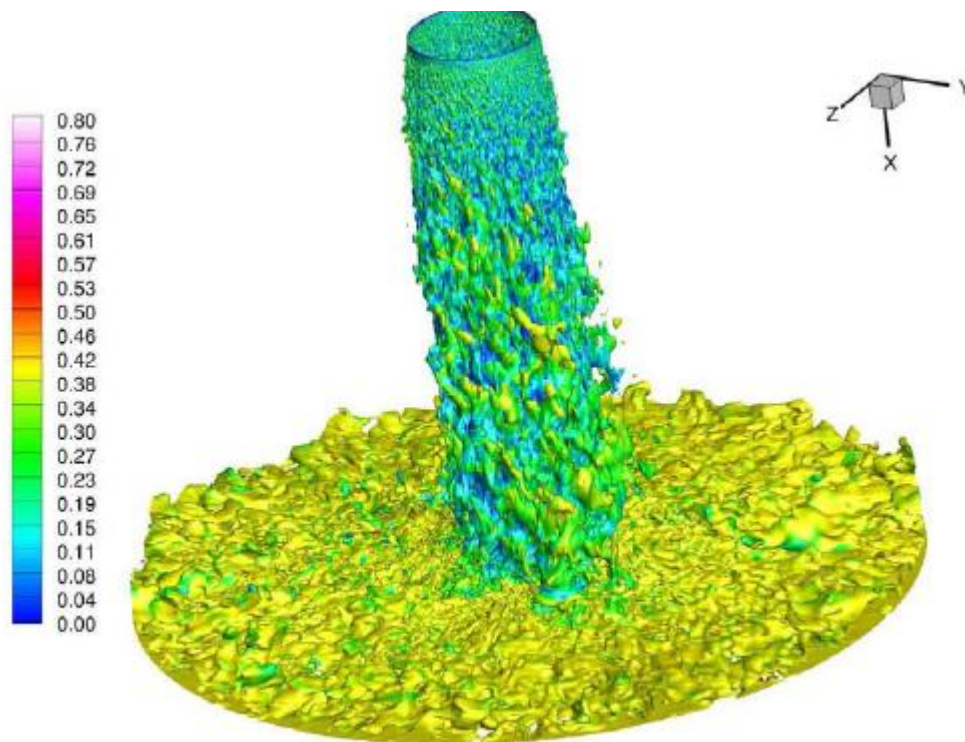
Alvi et al. - *AIAA J*, 2003, 2006; *JFM*:2008
Kumar et al. *AIAA J* 2009



Heated Mach 1.5 Jet Mean Axial Velocity (U/U_i)

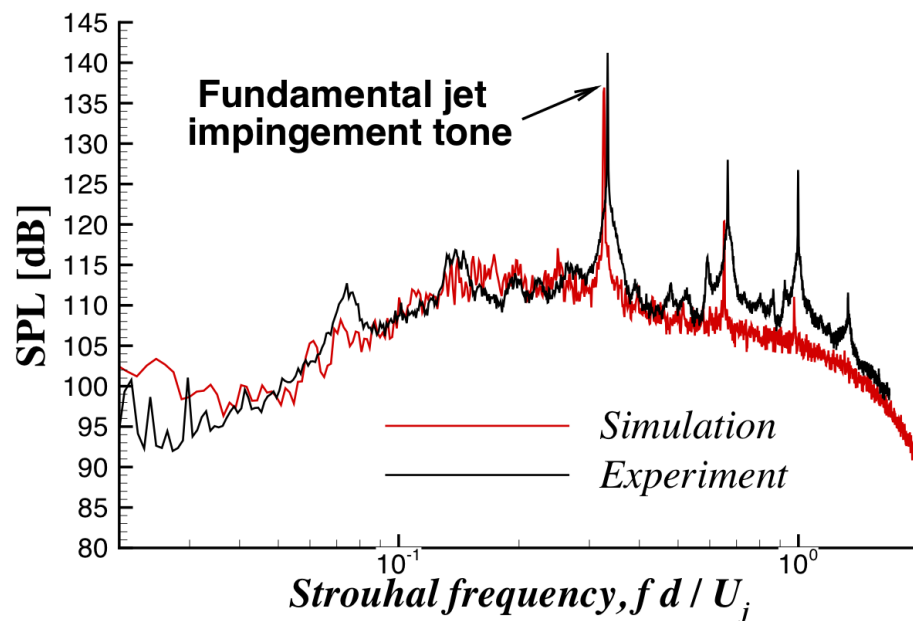
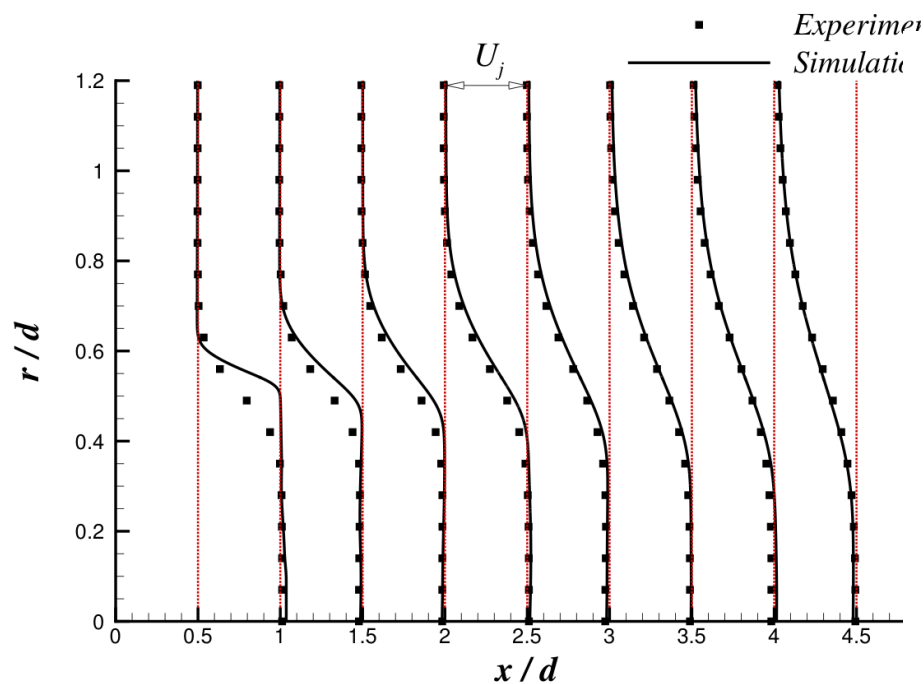


Iso-thermal Jet Total velocity magnitude iso-surfaces

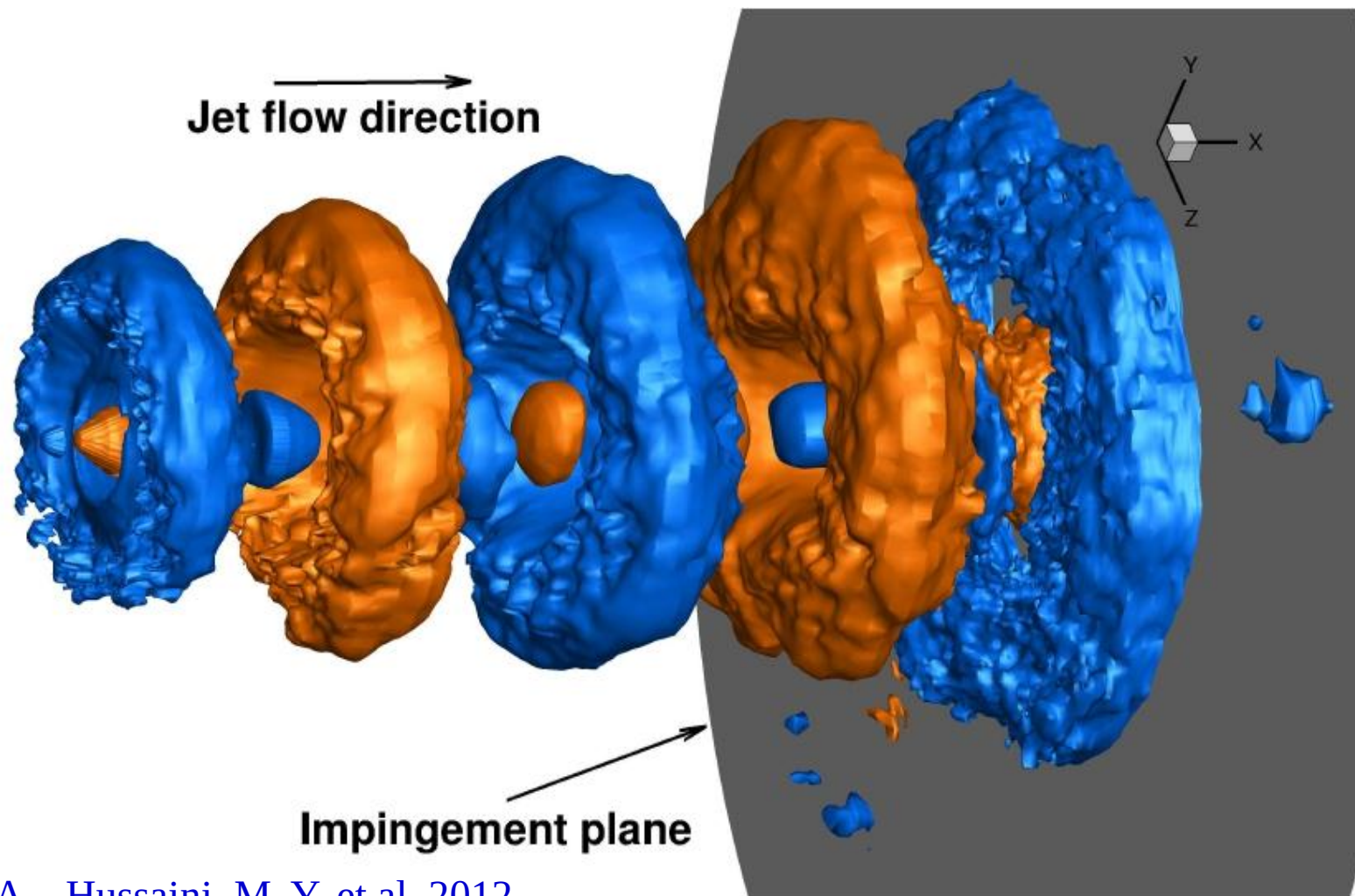


*Uzun, A. Hussaini, M. Y. et al, 2010

Heated Mach 1.5 Jet

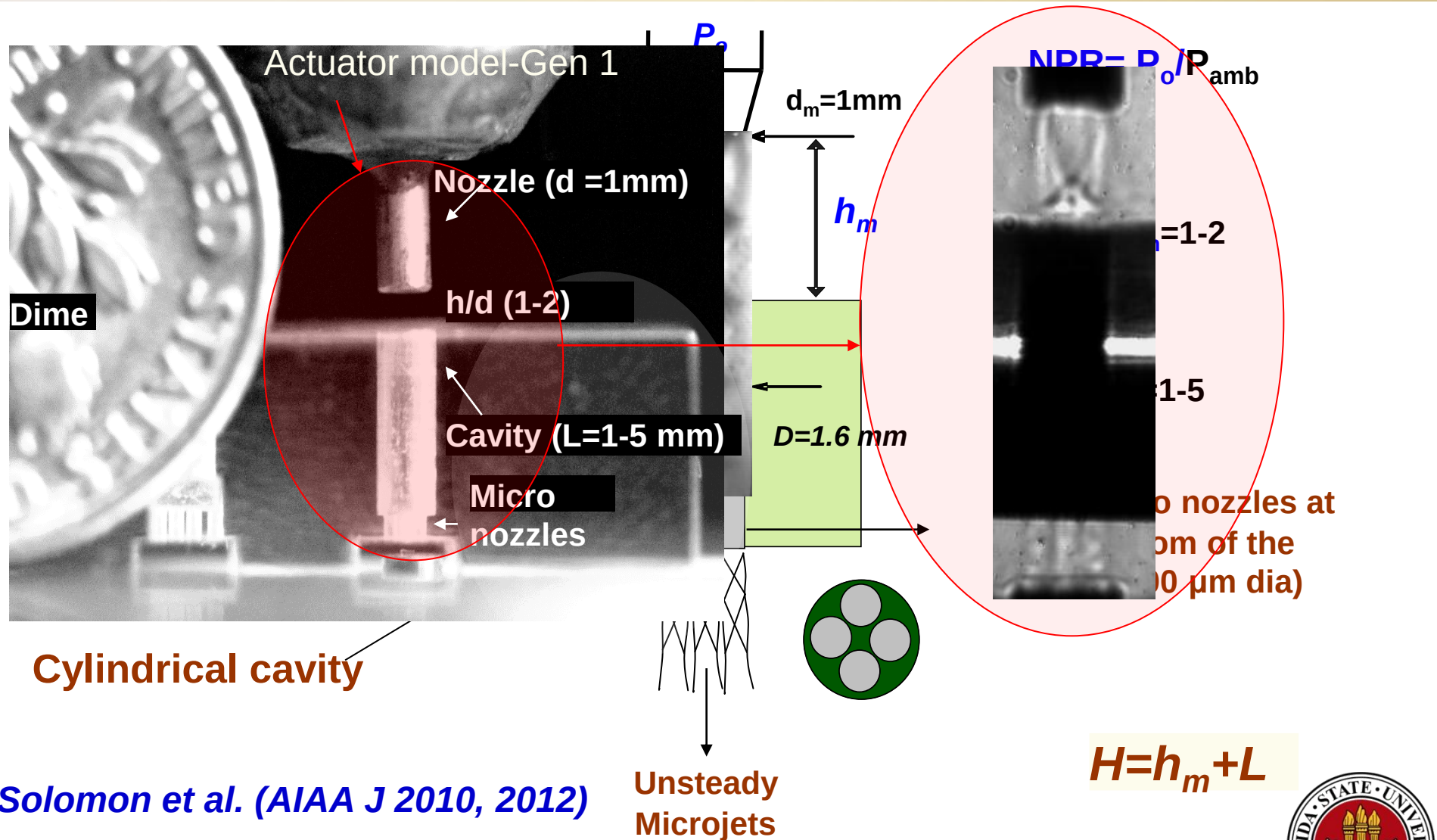


Pressure Iso-Surfaces Associated with Vortex Ring Structures at the Most Amplified Frequency using DMD

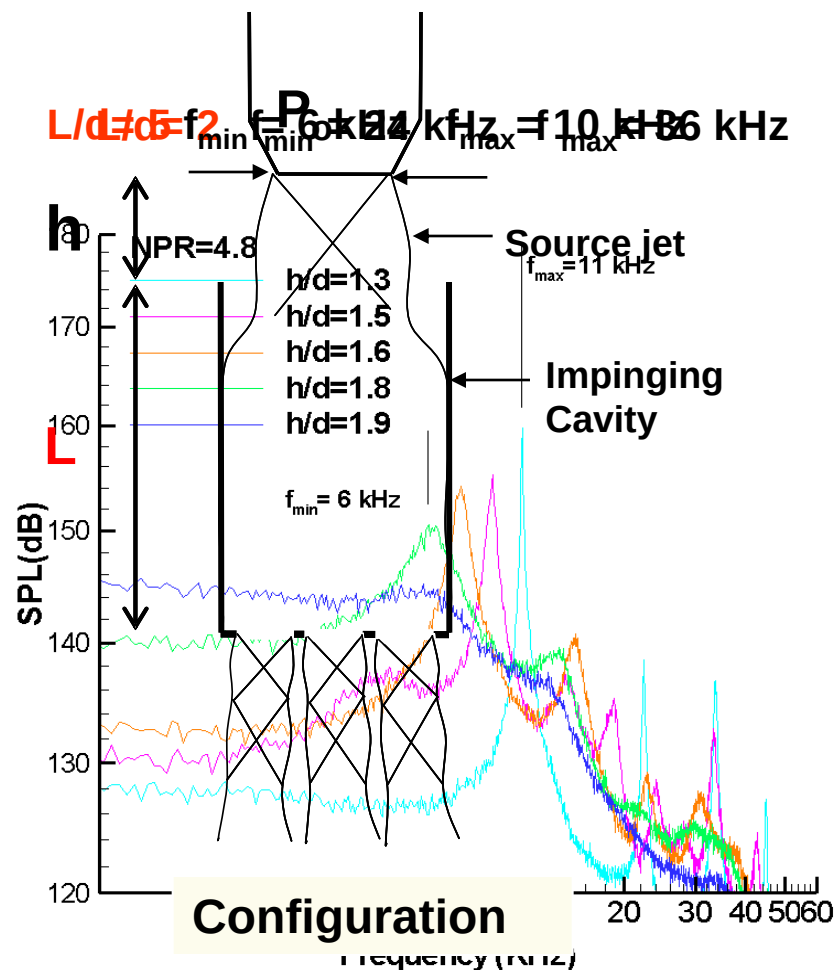
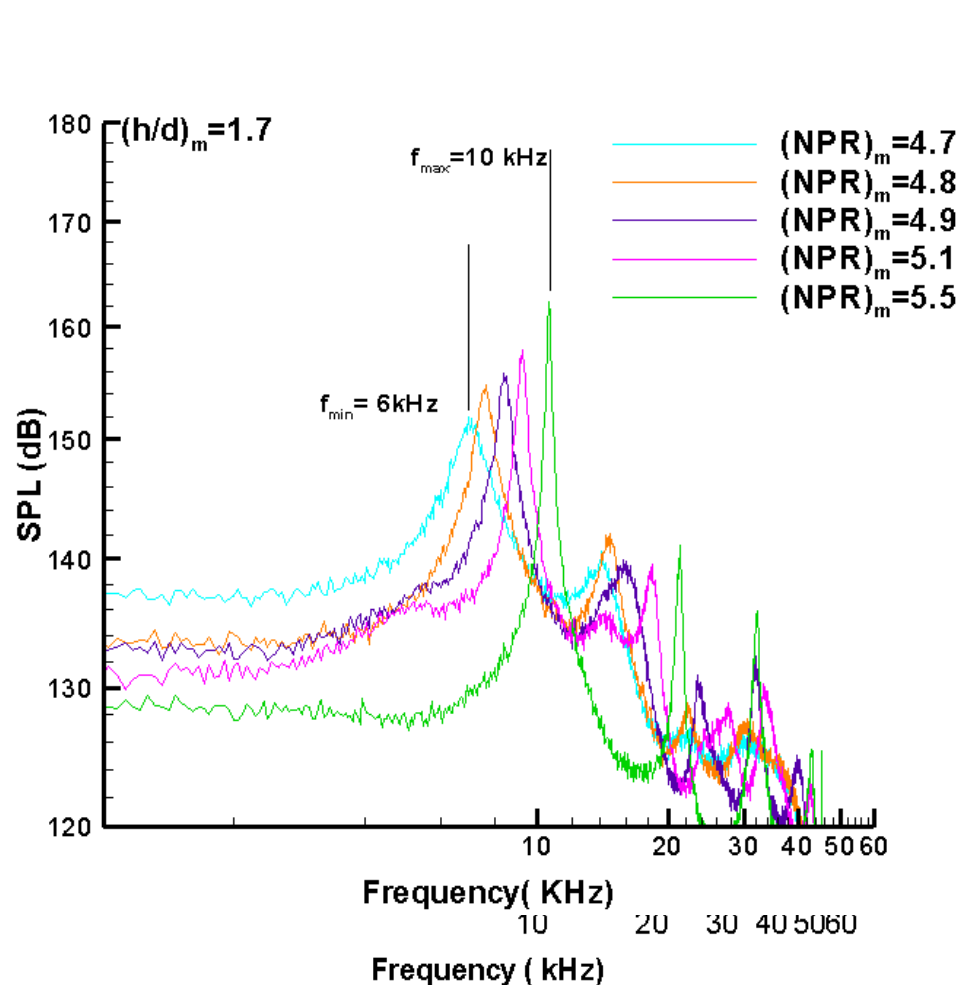


Uzun, A., Hussaini, M. Y. et al, 2012

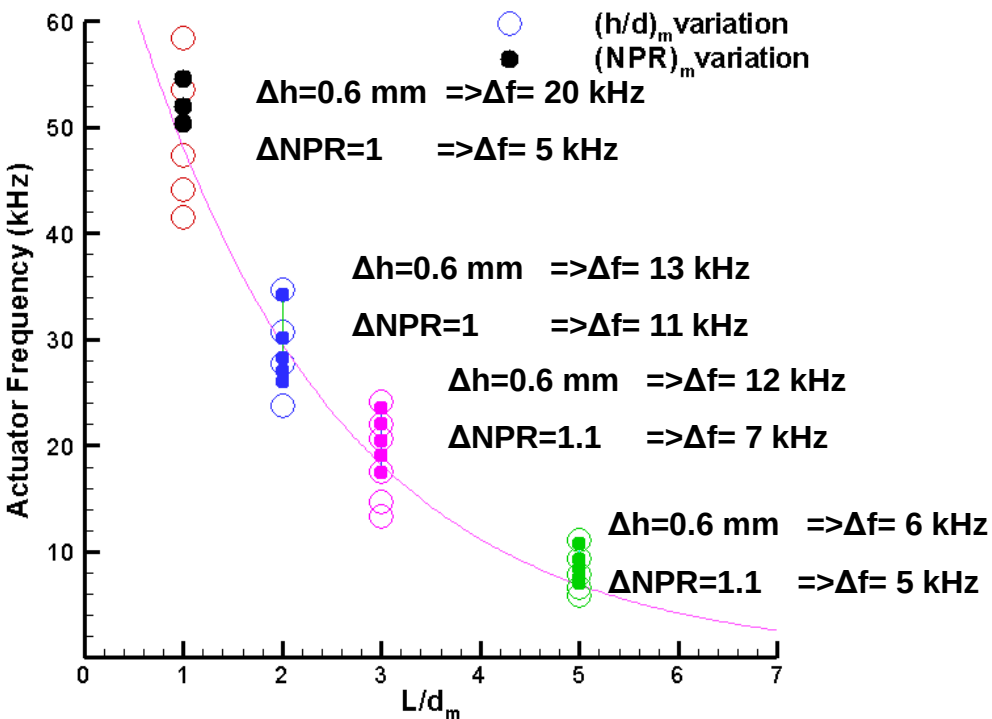
Pulsed (Micro)Actuators



$\Delta NPR \sim 1$ changes the frequency from 6-10 kHz



Actuator Performance Summary

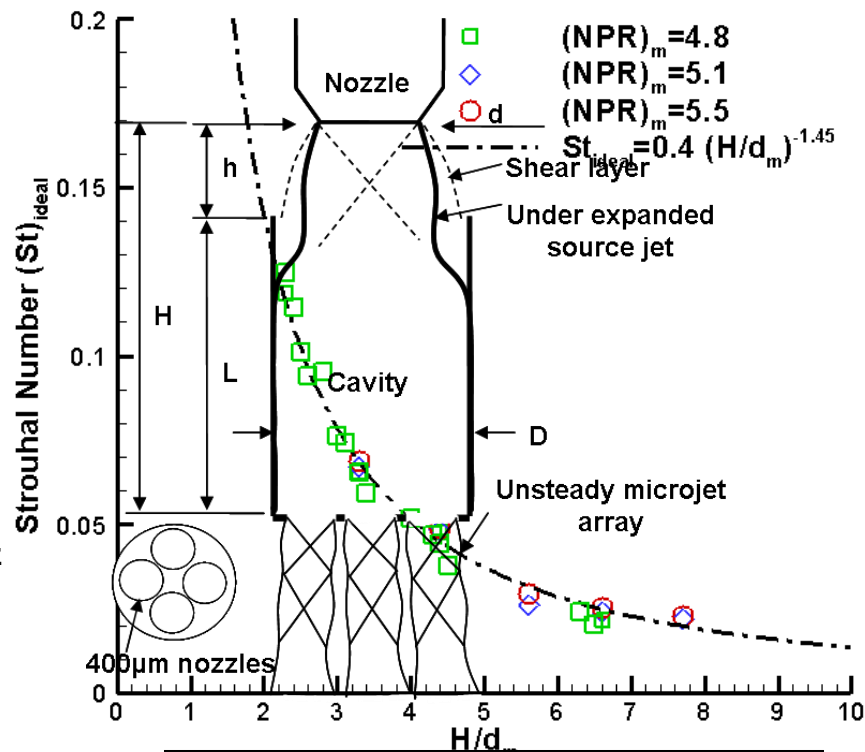


$$St_{ideal} = f d_m / U_{ideal}$$

d_m : diameter of source jet

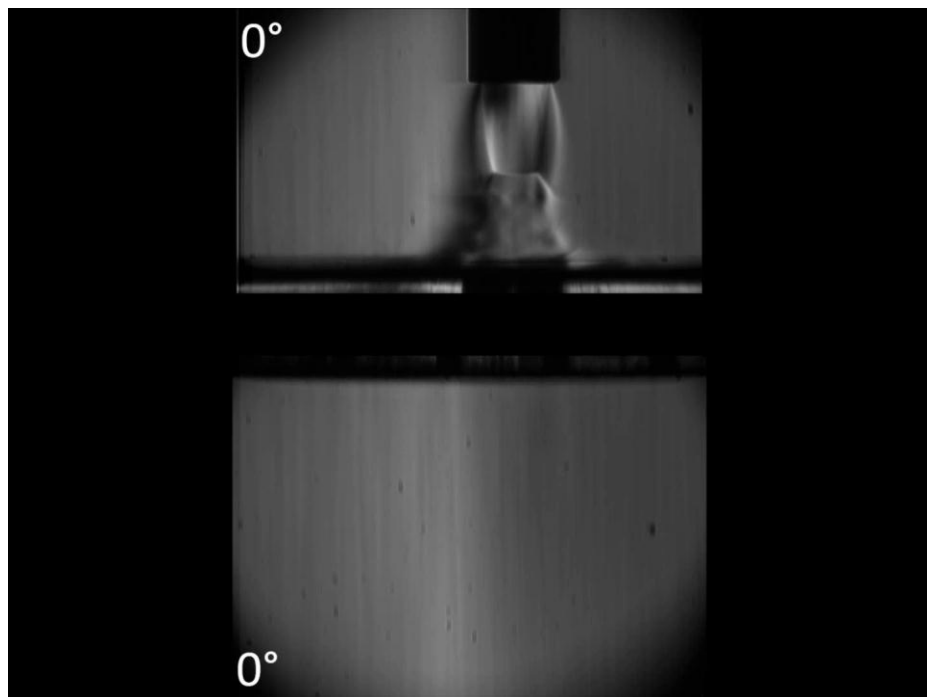
U_{ideal} : ideally expanded velocity corresponds to the source jet NPR

$$f = \frac{0.4}{d} \left(\frac{2}{\gamma - 1} \left((\text{NPR})^{\frac{\gamma-1}{\gamma}} - 1 \right) \right)^{1/2} \sqrt{\gamma R T_o (\text{NPR})^{\frac{1-\gamma}{\gamma}} (H/d)^{-1.45}}$$

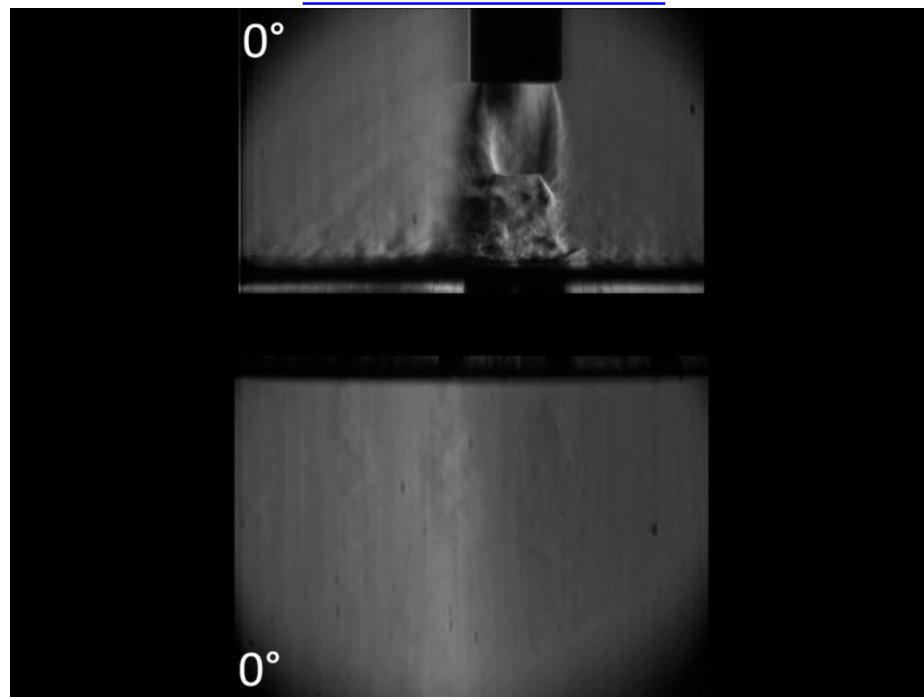


$$St_{ideal} = 0.4 (H/d_m)^{-1.45}$$

Phase Averaged



Instantaneous



100 images averaged at each phase

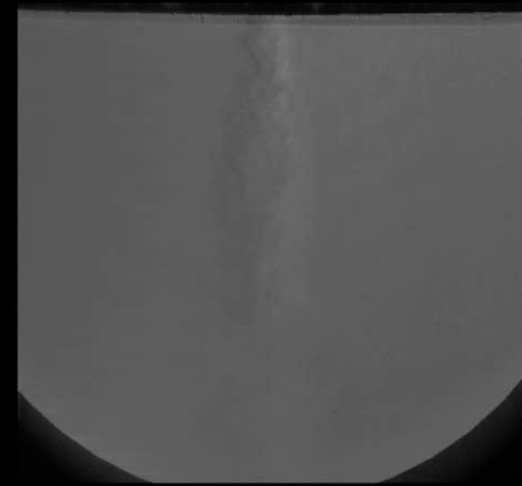
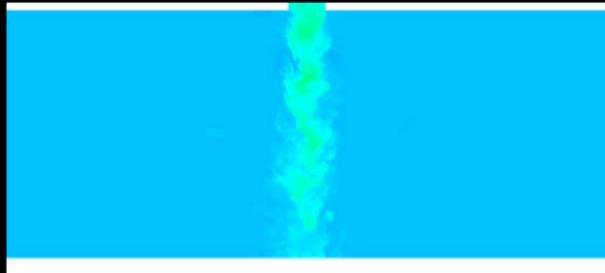
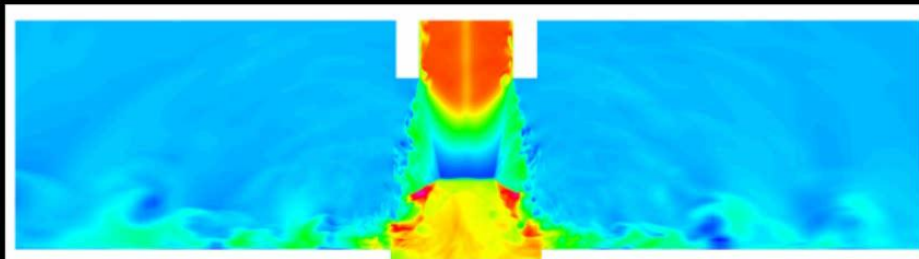
30° - 210° 'filling'*

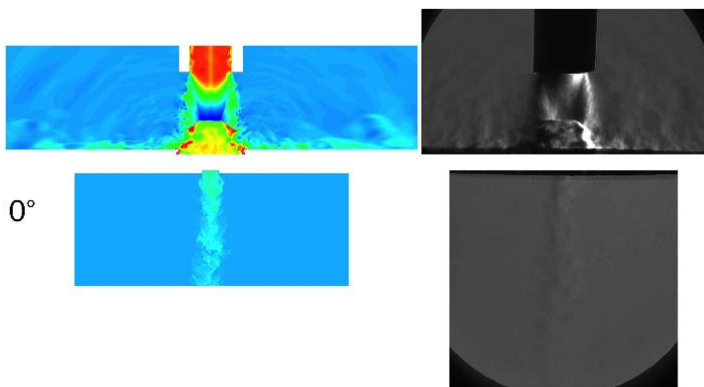
240° - 0° 'spilling'*

Foster, Alvi et al.
AIAA 2011

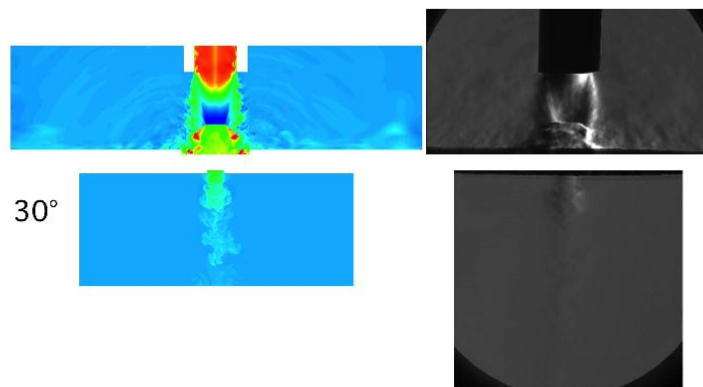


REM Actuator Simulations & Experiments

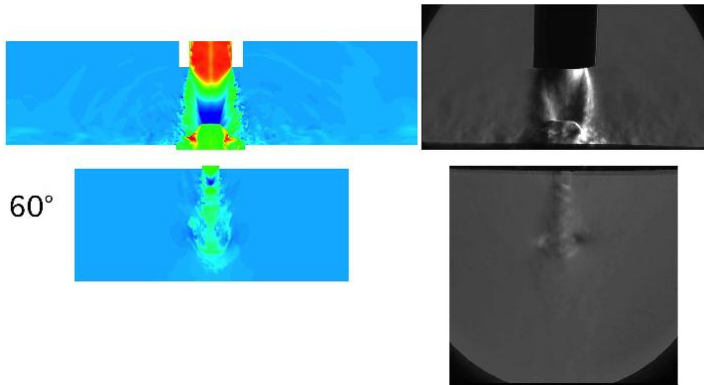




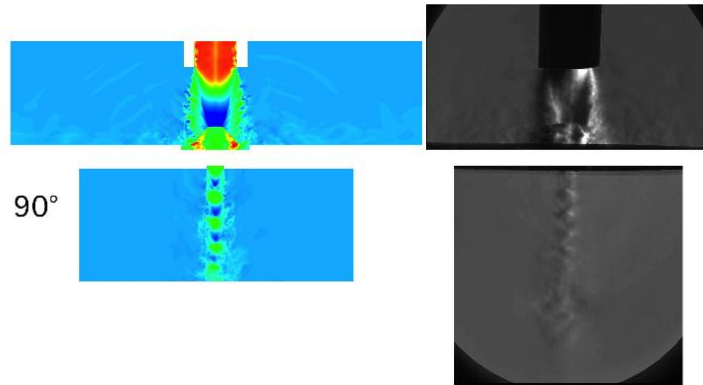
(a) Phase angle = 0°



(b) Phase angle = 30°



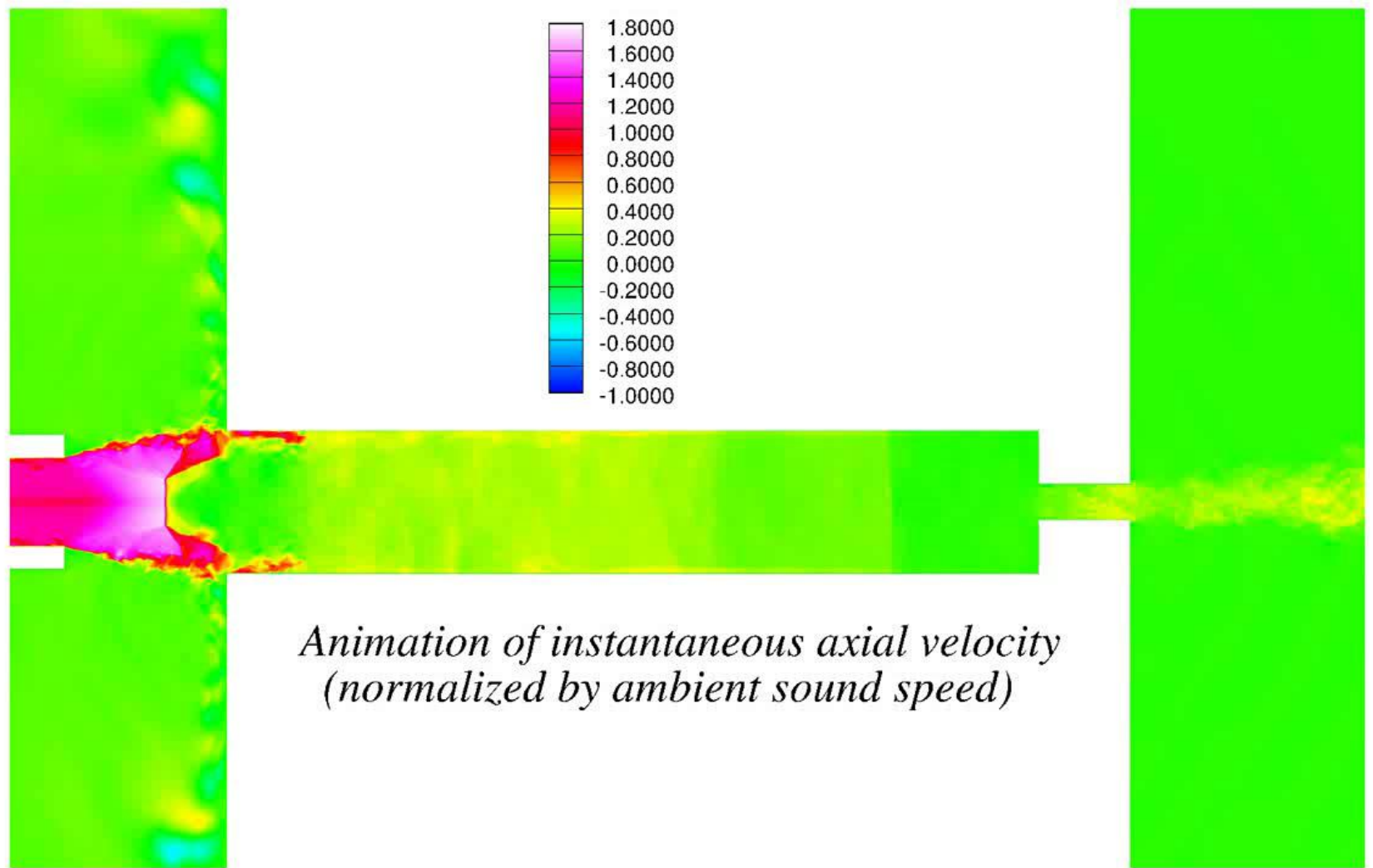
(c) Phase angle = 60°



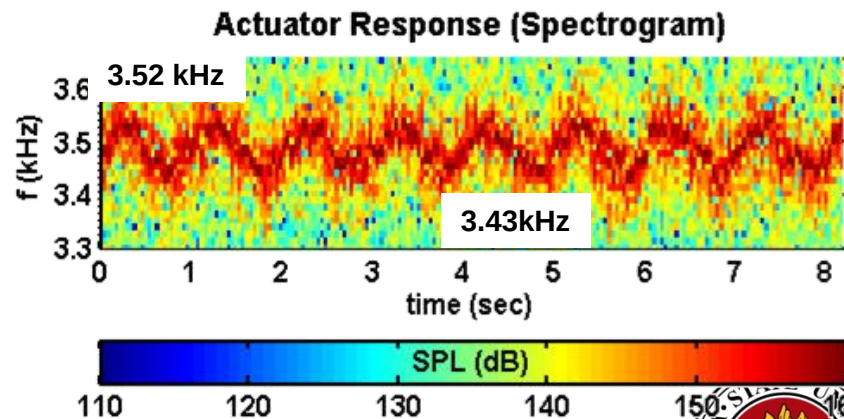
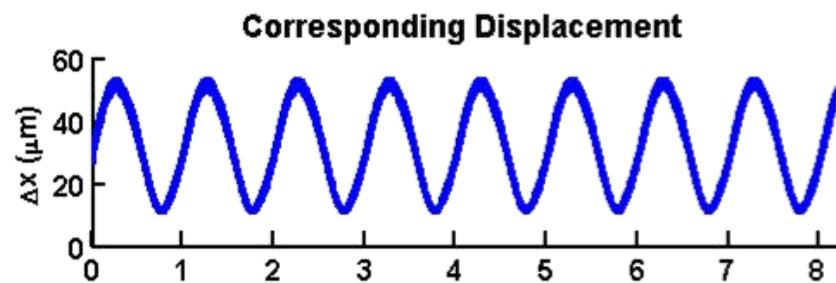
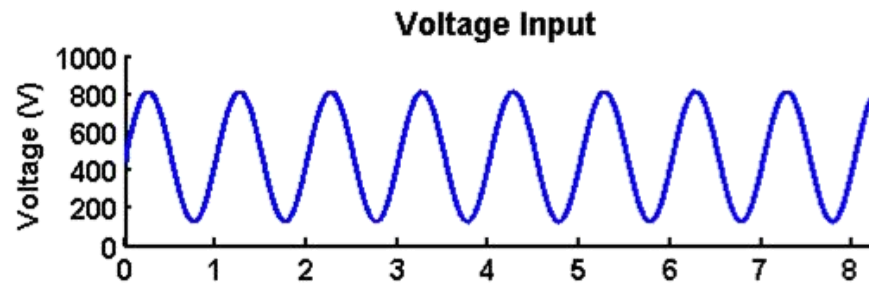
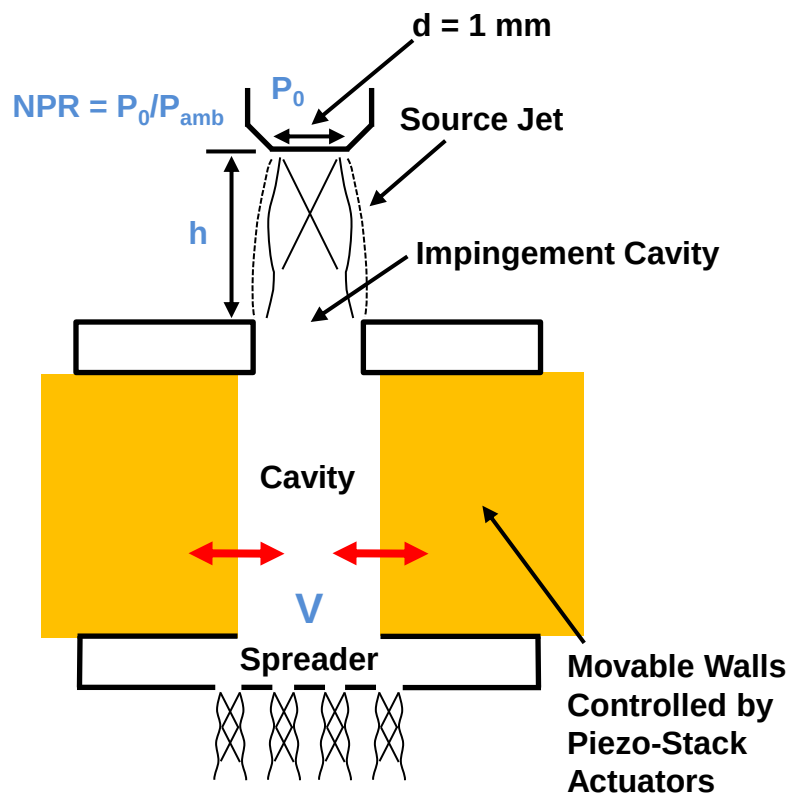
(d) Phase angle = 90°



Color map represents normalized density, $\rho/\rho_{\text{ambient}}$



SmartREM Actuator

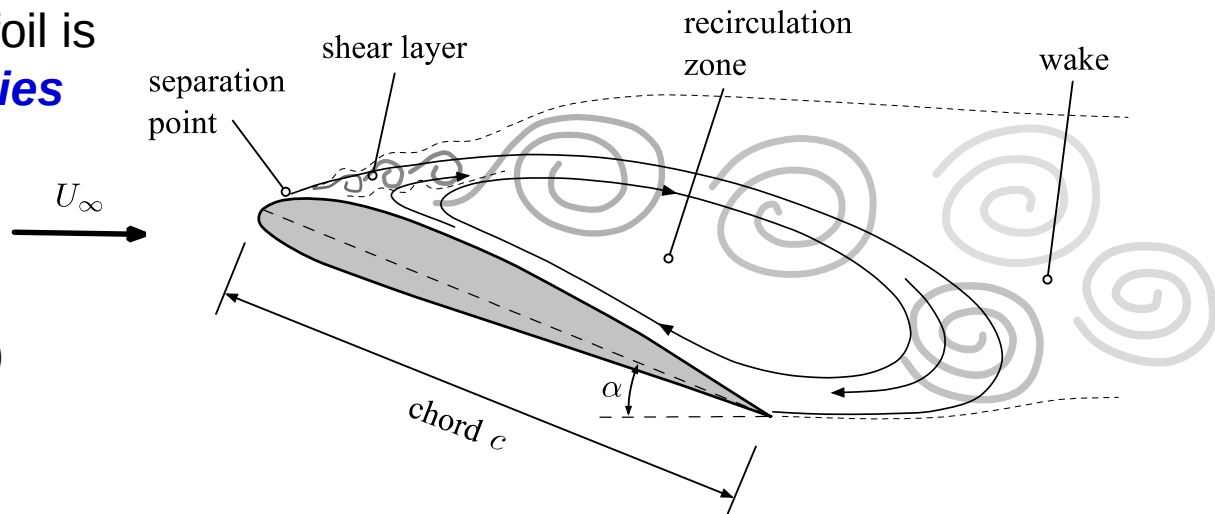


8/20/2012

Separated Flows (& Control of)*

- Separated flow past an airfoil is characterized by **frequencies associated with the**

- wake
- shear layer (SL)
- separation bubble (SB)
- actuation (if applied)



- “Lock-on” describes when these components are the same or are harmonics
 - Harmonics could be evidence of non-linear interactions
- Effectiveness of a control strategy may be related to the presence of lock-on** [Kotapati et al. 2001]

* Courtesy: Cattafesta, Mittal & Rowley

Experiments

- Increasingly sophisticated, providing high-fidelity data
 - 2-D/Stereo/Tomo-PIV /(Plenoptic)
 - => 2 component/3component/volumetric measurements
 - High-speed/time resolved and Phase conditioned
 - Synchronous: P , V , ρ ...
- They provide significant physical insight into flow physics.
- **Difficult and expensive to run; limited conditions**

Actuators

- A wider array of actuators with a range of control authority and bandwidth
 - Plasma Based (LAFPA, SJA, DBD); ZNMF (Synthtetic Jets); COMPAct, REM and more
 - **REM:** Simple, robust, scale-adapt/able, appropriate complexity & capability
 - *Smart* REM: ‘on-the-fly’ frequency control BUT more complex
- **Still unclear: “which actuator and under what conditions”**



Simulations

- Increasingly sophisticated, provide high-fidelity data for increasingly complex flows
 - More rigorous validation using better experimental data
 - Good to excellent agreement (for some cases?)
- Provide physical insight into flow physics, provide properties not easily measured.
- **Difficult and expensive to run; limited conditions**
- **Rarely go beyond experimental conditions?**

As we Progress

- Simulations+Theory used to better explore flow dynamics; larger range of conditions
- Provide guidance for active-adaptive control (that is *realistic/feasible*):
 - **Temporal and Spatial** requirements (freq., wavelength) for actuation and
 - **Location** (*where to best place them*)
 - **Type** of actuation: momentum, body force, thermal...
- Provide guidance for (sparse/minimal, practical) sensing requirements
- Help develop, simpler/low-order/..., *practical* models for closed-loop control
- **Plan experiments and computations together from the start**
 - We need to improve our communication skills

